## Chapter

Content

#### Gravitational field

- Gravitational force between point masses
- Gravitational field of a point mass
- Gravitational field near to the surface of the Earth
- Gravitational potential
- Circular orbits

#### Learning Outcomes

Candidates should be able to:

- (a) show an understanding of the concept of a gravitational field as an example of field of force and define gravitational field strength at a point as the gravitational force exerted per unit mass placed at that point.
- (b) recognize the analogy between certain qualitative and quantitative aspects of gravitational and electric fields (to be taught in Year 6).
- (c) recall and use Newton's law of gravitation in the form  $F = \frac{Gm_1m_2}{r^2}$ .
- (d) derive, from Newton's law of gravitation and the definition of gravitational field strength, the equation  $g = \frac{GM}{r^2}$  for the gravitational field strength of a point mass.
- (e) recall and apply the equation  $g = \frac{GM}{r^2}$  for the gravitational field strength of a point mass to new situations or to solve related problems.
- (f) show an understanding that near the surface of the Earth g is approximately constant and equal to the acceleration of free fall.
- (g) define the gravitational potential at a point as the work done per unit mass in bringing a small test mass from infinity to that point.

(h) solve problems using the equation  $\phi = -\frac{GM}{r}$  for the gravitational potential in the field of a point mass.

- (i) analyse circular orbits in inverse square law fields by relating the gravitational force to the centripetal acceleration it causes.
- (j) show an understanding of geostationary orbits and their application.

## 7.1 Gravitational Force F Introduction In 1687, Newton proposed in his Philosophiæ Naturalis Principia Mathematica that every

In 1687, Newton proposed in his *Philosophiæ Naturalis Principia Mathematica* that every mass attracts another mass with a force of gravity. According to him, two seemingly unrelated phenomena – the fall of an apple from an apple tree and the orbital motion of the planets around the Sun – are due to the same reason: gravitational attraction. He came up with the Newton's Law of Gravitation. This law is universally valid and applies to any planets in the solar system and even between distant galaxies.



## Newton's Law of Gravitation

Definition

Newton's law of gravitation states that two point masses attract each other with a force that is directly proportional to the product of their masses and inversely proportional to the square of the distance between them.

The magnitude of the gravitational force F between two particles of masses M and m which are separated by a distance r is given by



$$F=\frac{GMm}{r^2},$$

where G is the gravitational constant with a value of  $6.67 \times 10^{-11}$  N m<sup>2</sup> kg<sup>-2</sup>.

Sconstants can have units

#### Note:

- Newton's law of gravitation is an example of an inverse square law because the magnitude of the force varies inversely with the square of the separation of the particles.
- Gravitational force is attractive in nature and some books indicate this with a negative sign in the formula

$$Fd\frac{1}{r^2}$$

Example 7.1

A man of mass 85.0 kg is standing on the surface of the Earth. Given that the Earth has a mass of  $5.98 \times 10^{24}$  kg and a radius of  $6.37 \times 10^{5}$  m, calculate the force that the Earth exerts on the man. Deduce the force that the man exerts on Earth.

Solution  

$$F = \frac{GM_{E}m}{r_{E}^{2}}$$

$$= \frac{(6.67 \times 10^{-11}) \times (5.98 \times 10^{24}) \times (85.0)}{(6.37 \times 10^{6})^{2}}$$

$$= 836 \text{ N}$$

By Newton's third law, the force that the man exerts on the Earth is equal in magnitude to the force that the Earth exerts on the man, but in the opposite direction.

#### Note:

Due the mass of the Earth, we do not expect to see the Earth move:

$$F_{\text{man on Earth}} = -F_{\text{Earth on man}} \Rightarrow m_{\text{Earth}} a_{\text{Earth}} = -m_{\text{man}} a_{\text{man}} \Rightarrow a_{\text{Earth}} = -\frac{m_{\text{man}}}{m_{\text{Earth}}} a_{\text{man}} \approx 0$$

#### Example 7.2 NJC/H2/Prelims 2010/P1/16 - modified]

On the surface of the Earth, the gravitational force acting on an object is 45 N. When the object is at a height h above the surface, the gravitational force acting on it is 5 N. If R is the radius of the earth, calculate the approximate value for h in terms of R.

#### Solution

$$F_{surface} = \frac{G m m}{R^2} = 45$$

$$F_{1} = \frac{G m m}{R^2} = 45$$

$$F_{1} = \frac{G m m}{R^2} = 45$$

$$F_{1} = \frac{G m m}{R^2} = 45$$

$$F_{2} = \frac{G m m}{(R+h)^2} = 5$$

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$$F_{1} = \frac{G m m}{(R+h)^2} = 5$$

$$F_{1} = \frac{G m m}{(R+h)^2} = 6$$

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7.2	Gravitational Field Strength g
Gravitational Field Definition	A gravitational field is a region of space in which a mass placed in that region experiences a gravitational force.
Gravitational Field Lines	A region of gravitational field can be visualised as consisting of an array of imaginary field lines in which the tangent at a point on a field line acts in the direction of the gravitational force.
only althactive	<ul> <li>The direction of the field lines indicates the direction of the gravitational field.</li> <li>The density of the field lines indicates its strength. A region with a stronger gravitational field strength will have closer or denser field lines, and vice versa.</li> <li>For a point mass or a uniform spherical mass, the field lines are directed towards its centre.</li> <li>The figure on the left shows the field lines around Earth, which are directed towards its centre. If we zoom in to a region near the surface of the Earth, the field lines sector be parallel to one another as shown in the figure on the right.</li> <li>Hence, near the surface of the Earth, we can consider the gravitational field to be uniform with a magnitude of 9.81 N kg<sup>-1</sup>.</li> </ul>
and rever rep	utrive : duays into
Definition	The gravitational field strength at a point in space is defined as the gravitational force experienced per unit mass at that point.
Formula	$g = \frac{F}{m}$
Gravitational Field Strength of a Point	Since the gravitational force between two point masses is given by $F = \frac{GMm}{r^2}$ ,
Formula	$g = \frac{GM}{r^2}$
	<ul> <li>Gravitational field strength is a vector quantity and it points towards the mass which created it.</li> <li>The gravitational field strength at a point is the acceleration of free fall at that point.</li> <li>The resultant field strength at a point due to more than one mass can be found by the</li> </ul>
	vector sum of the gravitational field strengths due to all masses. * t/- defending acceleration due to frets Bee Pall
	Page 14 of 19 Dravitational field
	kg



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Variation in Gravitational Field Strength due to Earth's Rotation The figure shows an object of mass *m* hanging from a spring balance at two places on the Earth – the Equator and the North Pole.

In both cases there are two forces acting on the mass – gravitational force  $F_g$  (true weight) and force T due to the spring.



If the Earth is taken to be a uniform sphere, then  $F_g$  is the same at both the equatorial and polar regions. However, T will be different because the mass undergoes circular motion at the Equator.

#### Resultant force $F_{R}$ at the polar region:

- Applying Newton's second law, F<sub>R</sub> = F<sub>g</sub> T
- Since there is no circular motion, F<sub>R</sub>=0
- Therefore,  $T = F_g$
- The spring balance indicates F<sub>g</sub>, which is the true weight of the object.

#### Resultant force $F_R$ at the Equator:

- Applying Newton's second law,  $F_R = F_g T$
- Since the object undergoes circular motion,  $F_R = F_C = F_g T$
- Therefore,  $T = F_g F_c$
- The spring balance indicates a value less than the true weight of the object. This
  value is known as the apparent weight.

Important Note At the Equator, part of the gravitational force provides for the centripetal force to keep the object moving in a uniform circular motion.

Question: What happens if the Earth spins faster about its axis of rotation?



(i) Given that the Earth's radius is 6370 km, find the centripetal acceleration of an object placed at the Equator.

(ii) If a 5.0 kg mass is placed on a weighing scale at the Equator, what would the scale read in kg?

## $\frac{\text{Solution}}{\text{i}} = \frac{\sqrt{2}}{1 + (310 - 000)} + \frac{\sqrt{2}}{1 + (310 - 100)} + \frac{\sqrt{2}}{1 + (310 - 100)$

Other Factors Affecting Gravitational Field Strength

1.

The gravitational field strength over the Earth's surface varies as the result of its lack of spherical symmetry. Its polar diameter is about 40 km less than its equatorial diameter.



2. The Earth's density is not uniform. There are local variations in the Earth's gravitational field caused by mountains and trenches, as well as the presence of minerals and oil deposits.



shape





**Example 7.6** Given that the mass of the Earth is  $5.98 \times 10^{24}$  kg and its radius is  $6.37 \times 10^{6}$  m, calculate the change in the gravitational potential as an object is moved from the surface of the Earth to a point 1200 m above the surface.

$$\frac{\text{Solution}}{\Delta \phi} = \left(-\frac{Gm}{r}\right) - \left(-\frac{Gm}{r_E}\right)$$
$$= Gm \left(\frac{1}{r_E} - \frac{1}{r}\right)$$
$$= 6.67 \times 10^{-11} \times 5.98 \times 10^{24} \left(\frac{1}{6.37 \times 10^6} - \frac{1}{6.37 \times 10^6 + 1200}\right)$$
$$= 1.18 \times 10^4 \text{ d kg}^{-1}$$

Example 7.7 [SAJC/H2/Prelims 2010/P1/13 - modified] The figure shows two points X and Y at distances L and 2L from the centre of the Ear The gravitational potential at X is –8 kJ kg<sup>-1</sup>.



What is the work done when a 2 kg mass is taken from X to Y?

Solution  

$$\frac{\text{Solution}}{\phi_{p} \cdot \phi_{i}^{+} + \Delta \phi} \quad \phi_{x} \cdot -\frac{G_{m}}{L} \cdot -8 \text{ Fkg}^{-1} \\
= -\frac{G_{m}}{TL} \quad \phi_{y}^{+} \cdot -\frac{G_{m}}{2L} \cdot -4 \text{ Fkg}^{-1} \\
\text{work done' gain in GPE \cdot \Delta u} \\
: m \Delta \phi \quad m (\phi_{y} - \phi_{x}) \\
\cdot 2 \left[ (-2) \cdot (-8) \right] \\
\cdot 8 \text{ kJ}$$

#### 7.4

#### Escape Velocity

Have you ever wondered if it is possible to throw an object into the air at such a high velocity that it will never fall back to Earth? This is actually plausible because the acceleration of free fall does not stay at 9.81 m s<sup>-2</sup> but decreases as the object moves away from the Earth. If this is possible, what is the value of this "escape velocity"? The solution to this problem is simple if we consider the energy of the object.

How much energy does an object of mass *m* need in order to escape from the Earth's gravitational influence and reach infinity? At infinity, its GPE is defined as zero. If the object has sufficient energy to *just* reach infinity, its KE at infinity is zero. So the total energy at infinity is given by  $E_T = E_P + E_K = 0$ .

From the principle of conservation of energy, the total energy of an object should remain unchanged throughout its motion. This means that any object with a total energy zero will be able to *just* reach infinity and *stop* there.

Consider an object (e.g. a rocket) of mass m being projected with a velocity v from the surface of the Earth. Its initial GPE and KE are:

$$U = -\frac{GMm}{R}$$
 and  $KE = \frac{1}{2}mv^2$ 

Therefore, minimum energy to reach infinity is such that

Total energy 
$$= -\frac{GMm}{R} + \frac{1}{2}mv^2 = 0$$
  
 $v = \sqrt{\frac{2GM}{R}} = \sqrt{2gR}$ 
(a) suming conservation of energy)

Note:

Formula

Escape Velocity

- The escape velocity is independent of the mass of the object.
- The escape velocity for Earth is 11.2 km s<sup>-1</sup>.

**Example 7.8** A spacecraft of mass 5000 kg propels itself from the surface of the Earth of mass  $5.98 \times 10^{24}$  kg and radius  $6.37 \times 10^6$  m. If Calculate the minimum escape speed.

Solution

$$v \ge \sqrt{\frac{2GM}{R}} \qquad v \ge \sqrt{2gR} \\ = \sqrt{\frac{2 \times (6.67 \times 10^{-11}) \times (5.98 \times 10^{24})}{6.37 \times 10^6}} \quad OR \qquad = \sqrt{2 \times 9.81 \times (6.37 \times 10^6)} \\ = 1.12 \times 10^4 \text{ m s}^{-1} \qquad = 1.12 \times 10^4 \text{ m s}^{-1}$$

#### Example 7.9 [RI/H2/Prelims 2010/P2/3 - part]

Given that the mass of Earth is  $5.98 \times 10^{24}$  kg and its radius is 6370 km, determine the minimum kinetic energy required to project a spacecraft of mass 2550 kg from the surface of Earth so that it completely escapes from the gravitational attraction of the Earth. Ignore air resistance.

#### Solution

By the principle of conservation of energy, total energy on surface = total energy at infinity

At infinity, GPE is 0 (by definition) and KE is 0 (since spacecraft is launched with the minimum kinetic energy).

Hence,

$$E_{surface} = E_{infinity}$$

$$f_{Tur} \mathcal{K} = \int_{k_{min}} \left( -\frac{Gm_{E}m_{S}}{R_{E}} \right) \cdot \delta$$

$$E_{k_{min}} = \frac{Gm_{E}m_{S}}{R_{E}}$$

$$, \left( \frac{G(R_{E}m_{S})}{R_{E}} + \frac{Gm_{E}m_{S}}{R_{E}} \right) \cdot \frac{G(R_{E}m_{S})}{(S_{E}m_{S})} + \frac{G(R_{E}$$



Recap

For any object of mass m to move in uniform circular motion of radius r, there must be a resultant force  $F_c$  acting on the object which is directed towards the centre of the circle. This is known as the centripetal force, which can be expressed as

$$F_{\rm c} = \frac{mv^2}{r}$$
 or  $mr\omega^2$ 

where v is the linear velocity and  $\omega$  is the angular velocity.

Planetary & Satellite Motions

The time taken for the object to make one complete circle is known as the period T, which is given by

 $T=\frac{2\pi}{\omega}=\frac{2\pi r}{v}$ 

Kepler's Third Law Consider a satellite of mass m moving with linear speed v in a circular orbit of radius r about a planet of mass M.



The only force acting on the satellite is the gravitational force. Hence, this gravitational force provides the centripetal force necessary for the circular motion of the satellite. Satellites generally do not have any engine propelling it.

Applying Newton's second law of motion, (know thus)

$$F_{g} = F_{c}$$

$$G \frac{Mm}{r^{2}} = mr\omega^{2}$$

$$G \frac{Mm}{r^{2}} = mr\left(\frac{2\pi}{T}\right)^{2}$$

$$T^{2} = \frac{4\pi^{2}}{GM}r^{3}$$

$$T^{2} \propto r^{3}$$

where T is the orbital period and r is the orbital radius. This relationship is also known as Kepler's third law.

#### Example 7.10 [MI/H2/Prelims 2010/P1/14 - modified]

Mercury is  $5.79 \times 10^{10}$  m from the Sun and it takes 0.241 Earth years for Mercury to make one revolution around the Sun. If Neptune is  $450 \times 10^{10}$  m from the Sun, calculate the period of Neptune's around the Sun.

Solution

1.

$$\int \frac{4\pi^{2}}{Gm} r^{3},$$

$$\left(\frac{T_{N}}{T_{m}}\right)^{2} = \left(\frac{r_{N}}{r_{M}}\right)^{3}$$

$$\left(\frac{T_{N}}{\sigma \cdot 2\pi i \ \gamma r}\right)^{2} = \left(\frac{45\sigma \times 10^{10}}{5 \cdot 3\pi 10^{10}}\right)^{3}$$

$$\frac{T_{N}}{\sigma \cdot 2\pi i \ \gamma r} = 1(5 \ \gamma \tan s)$$

p radius : rearts + altitude

**Example 7.11** (a) An object orbits at an altitude of 300 km above the Earth. What is its linear velocity?

(b) Briefly explain why the orbital plane of any satellite includes the centre of the Earth.

Take Earth as a uniform sphere of radius 6370 km and mass 5.98 × 10<sup>24</sup> kg.

Solution

(a) rbital speed: 
$$\frac{ugs}{tot}$$
 Newton's grd law  
 $G_{TT} = \frac{mV2}{r}$   
 $v = \int \frac{G_{TT}}{G_{T}}$   
 $\frac{(3\times10^{-11}\times5.95\times10^{24})}{(3706001306600)}$   
 $3\times10^{3}ms^{-1}$ 

e gravitational force due to the Earth provides for the centripetal force on the atellite. Since the centripetal force is directed towards the centre of the Earth, the orbital plane must contain the centre of the Earth.

special orbit "

A geostationary satellite is one that remains at a fixed position relative to any point on Geostationary Orbits the Earth's surface. A geostationary satellite must satisfy the following conditions: Important Note 1. Its orbital period is the same as that of the Earth about its axis of rotation (24 hrs). 2. Its direction of rotation is the same as that of the Earth about its axis of rotation (eastward). 3. It lies above the Equator (or in the same plane as the Equator). The radius of a geostationary orbit is  $r = \left(\frac{T^2 G M}{4\pi^2}\right)^{\frac{1}{3}} = \left(\frac{(24 \times 3600)^2 \times 6.672 \times 10^{-11} \times 5.96 \times 10^{24}}{4\pi^2}\right)^{\frac{1}{3}} = 42200 \text{ km}$ Therefore, the height of the orbit above the Earth's surface = 42200 - 6400 = 35800 km. Advantages of using a geostationary satellite 1. Continuous surveillance of the region under it. 2. Easy for the ground station to communicate with it. 3. Due to the high altitude, the satellite can transmit and receive signals over a large area. can it lie at any other latitudes ? Disadvantages Its distance from the Earth's surface is large compared to the Low-Earth Orbit (LEO) satellites, which typically operate at an altitude of a few hundred kilometres. This leads to 1. a significant loss of signal strengths, 2. poorer resolution in imaging satellites, 3. time lag in telecommunication. Which of the following quantities are not necessarily the same for satellites that are in Example 7.12 geostationary orbits around the Earth? A. angular velocity B. centripetal acceleration C. kinetic energy D. period of orbit E. radius of orbit F mass G. momentum Solution:

(, F, y. (any quantity that involves muss)

#### Example 7.13 [N99/III/2]

A satellite of mass 2400 kg is placed in a geostationary orbit at a distance of  $4.23 \times 10^7$  m from the centre of the Earth.

- (a) Calculate
  - 1. the angular velocity of the satellite,
  - 2. the speed of the satellite,
  - 3. the acceleration of the satellite,
  - 4. the force of attraction between the Earth and the satellite,
  - 5. the mass of the Earth.
- (b) Explain why a geostationary satellite
  - 1. must be placed vertically above the equator,
    - 2. must move from west to east.
- (c) Why are such satellites often used for telecommunications?

#### Solutions

(a) 1.	$\omega = \frac{2\pi}{T}$
	$=\frac{2\pi}{24\times60\times60}$
	$= 7.27 \times 10^{-5} \text{ rad s}^{-1}$
2.	$v = r\omega$
	$= 4.23 \times 10^7 \times 7.27 \times 10^{-5}$
	=3.08×10 <sup>3</sup> m s <sup>-1</sup>
3.	$a = \frac{v^2}{r}$
	$=\frac{(3.08\times10^3)^2}{4.23\times10^7}$
	= 0.224 m s <sup>-2</sup>
4.	$F = \frac{GMm}{r^2}$
	= <i>ma</i>
	= 2400 × 0.224
	= 538 N
5. 53	$b8 = \frac{GMm}{r^2}$
	$M = \frac{538 \times (4.23 \times 10^7)^2}{6.11 \times 10^{-11} \times 2400}$
	$= 6.56 \times 10^{24}$ kg

- (b) 1. Since the centripetal force on the satellite is directed towards the centre of the Earth, any circular orbit must have its centre at the centre of the Earth. If the orbit is not on the equator, the satellite must sometimes be over the northern hemisphere and sometimes over the southern hemisphere, and so cannot be geostationary.
  - 2. For a satellite to stay above a fixed point, it must have the same direction of rotation as the Earth about its axis of rotation, ie from west to east.
- (c) The earthbound transmitters and receivers can be aimed in a fixed direction, with no disruption to the signals.

# Gravitational force: $F = \frac{GMm}{r^2}$ Gravitational field strength: $g = \frac{GM}{r^2}$ Gravitational potential energy: $U = -\frac{GMm}{r}$ Gravitational potential: $\phi = -\frac{GM}{r}$ Relationship between F and g: $g = \frac{F}{m}$

Summary

• Relationship between *U* and  $\phi$ :  $\phi = \frac{U}{m}$ 

• Relationship between *F* and *U*: 
$$F = -\frac{dU}{dr}$$

• Relationship between g and  $\phi$ :  $g = -\frac{d\varphi}{dr}$  = negative potential gradient.

$$-\frac{GMm}{R_{Earth}} + \frac{1}{2}mv^{2} = 0 \implies \qquad v_{escape} = \sqrt{\frac{2GM}{R_{Earth}}} = \sqrt{2gR_{Earth}}$$

• For a body to orbit around the Earth at radius r.

$$\frac{GMm}{r^2} = \frac{mv^2}{r} \Rightarrow \qquad v_{orbital} = \sqrt{\frac{GM}{r}}$$
$$\frac{GMm}{r^2} = mr\left(\frac{2\pi}{T}\right)^2 \Rightarrow \qquad T^2 = \frac{4\pi^2}{GM}r^3$$

- These conditions will ensure that a geostationary satellite is always above the same point on the Earth's surface, ie its relative position with respect to the Earth remains unchanged:
  - 1. Same 24-hour orbital period as the Earth.
  - 2. Same Eastward rotation as the Earth.
  - 3. Orbits above the equator.



Consider a point P located within a solid sphere a distance *r* from the centre of the sphere. We will prove that the "effective mass" consists of only the mass enclosed by the shaded sphere within dotted sphere of radius *r*. The approach is to prove that the rest of the mass (the unshaded portion) cancels out each other's gravitational effect such that their net force on any mass placed at P is zero.

First, consider a thin spherical shell of thickness *L*. P experiences the gravitational field from every portion of the shell. Two pieces of mass A and B are located on the opposite sides of P as shown. If the angle they subtend at P is  $\theta$ , then their areas (considering a square of sides = arc length,  $r\theta$ ) are given by  $(r_A\theta)^2$  and  $(r_B\theta)^2$  respectively. So their masses will be  $(r_A\theta)^2 L\rho$  and  $(r_B\theta)^2 L\rho$  respectively, where  $\rho$  is the density of the sphere. Hence, their gravitational field strength at P will be  $G \frac{(r_A\theta)^2 L\rho}{r_A^2}$  and  $G \frac{(r_B\theta)^2 L\rho}{r_A^2}$  respectively.

In both cases the *r*'s cancel away and we get  $G\theta^2 L\rho$  for both. So the gravitational field strengths of A and B are equal in magnitude at P. It is also obvious that they are opposite in direction since A and B are on opposite sides of P. So the resultant field at P due to A and B is zero. Apply this argument to every direction around P, we find that the resultant field due to the whole spherical shell is zero.

A solid sphere can be imagined to be made up of layers of spherical shells. If every shell with radius > r produces zero resultant field at P then clearly their combined effect is still zero. So the effective mass is only those within the dotted shaded sphere.

Derivation of  $\Delta U = mgh$ 

Appendix **B** 

When an object of mass m is lifted through height h from point 1 to point 2 near the surface of the Earth, we usually use the formula mgh to determine the gain in gravitational potential energy U. This is a good approximation if g is constant between point 1 and point 2.

#### Proof

Take point 1 to be the ground level and point 2 a distance h above it, such that  $r_1 = R_E$  and  $r_2 = R_E + h$ . Then

$$\Delta U = \left(-\frac{GMm}{r_2}\right) - \left(-\frac{GMm}{r_1}\right)$$
$$= \left(-\frac{GMm}{R_E + h_2}\right) - \left(-\frac{GMm}{R_E}\right)$$
$$= -\frac{GMm}{R_E}\left(\frac{R_E}{R_E + h} - 1\right)$$
$$= -\frac{GMm}{R_E}\left(\frac{R_E - R_E - h}{R_E + h}\right)$$
$$= \frac{GMm}{R_E}\left(\frac{h}{R_E + h}\right)$$
$$\approx \frac{GMm}{R_E}\left(\frac{h}{R_E}\right) \qquad \text{since } h << R_E$$
$$= m\left(\frac{GM}{R_E^2}\right)h \qquad \text{since } g = \frac{GM}{R_E^2}$$

= mgh